

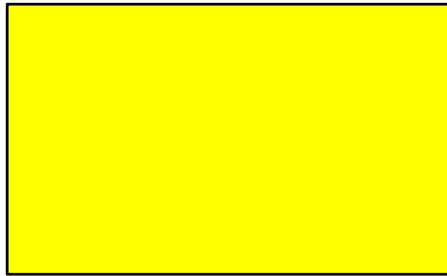
Workshop - MSC Nastran Topometry Optimization of a Composite Panel

AN MSC NASTRAN SOL 200 TUTORIAL

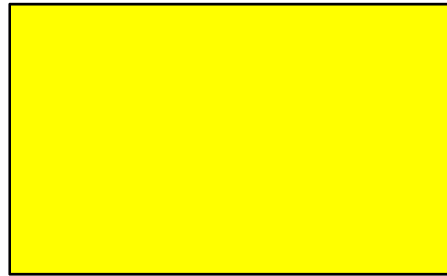
Goal: Use Nastran SOL 200 Optimization

Use Topometry optimization to determine ply shapes

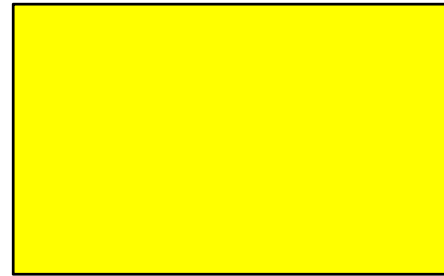
Initial Ply
Shapes



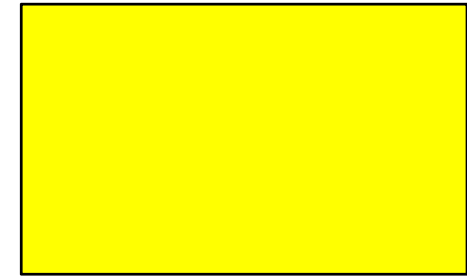
Ply 1 and 8
80°



Ply 2 and 7
-65°



Ply 3 and 6
80°

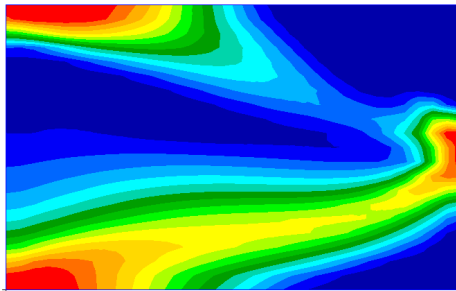


Ply 4 and 5
-65°

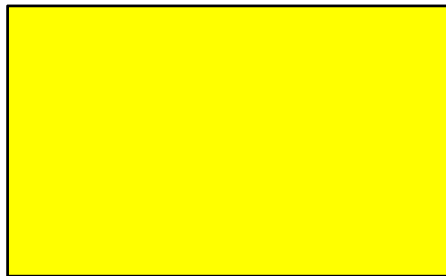
Goal: Use Nastran SOL 200 Optimization

Use Topometry optimization to determine optimal ply shapes

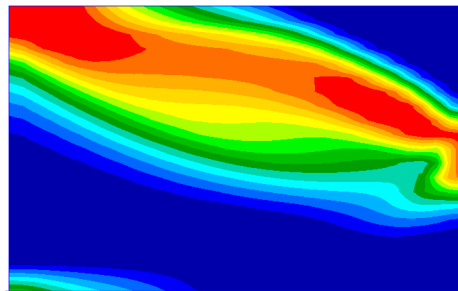
Topometry
Results



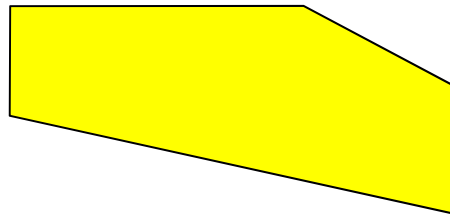
Ply 1 and 8
80°



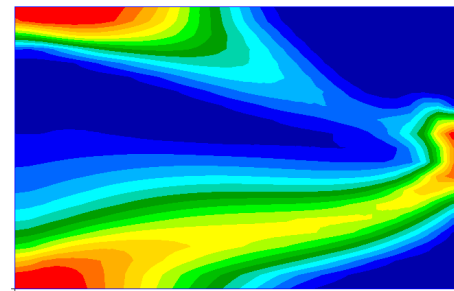
Ply 1 and 8
80°



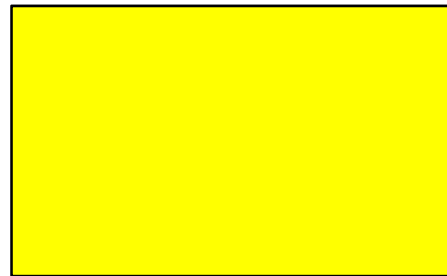
Ply 2 and 7
-65°



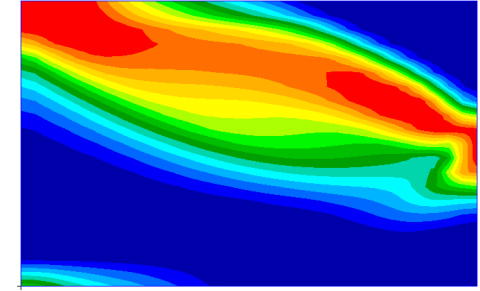
Ply 2 and 7
-65°



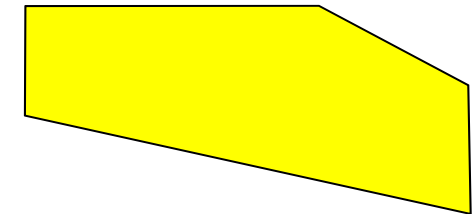
Ply 3 and 6
80°



Ply 3 and 6
80°



Ply 4 and 5
-65°



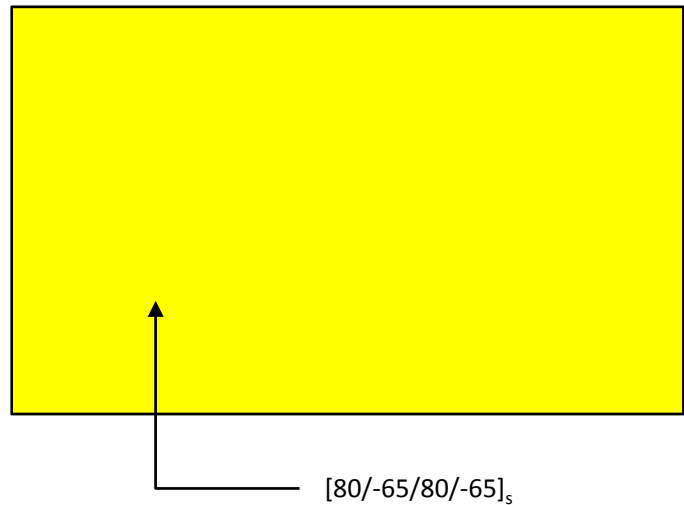
Ply 4 and 5
-65°

Proposed Ply
Shapes

Goal: Use Nastran SOL 200 Optimization

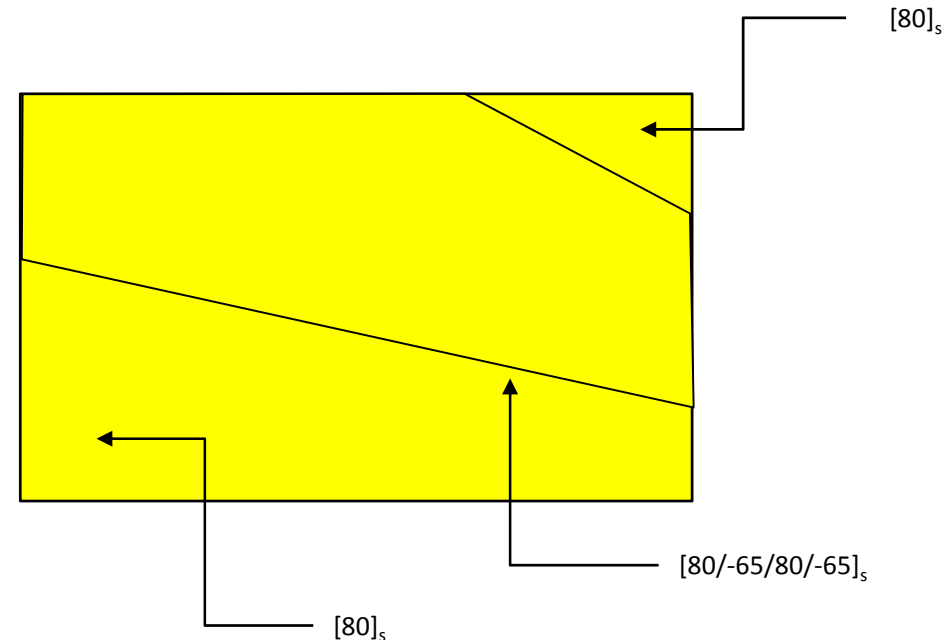
Before Optimization

- Panel of initial layup

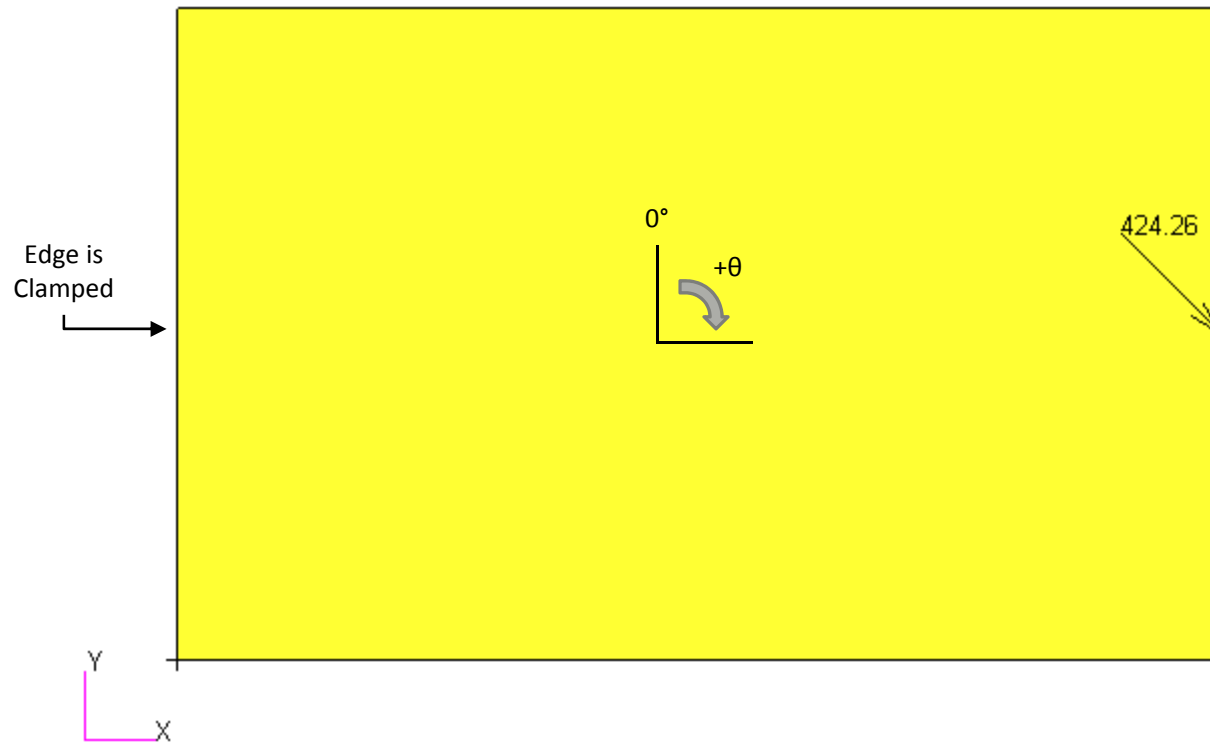


After Optimization

- Panel of updated layups
- The weight is reduced



Details of the structural model



Panel: 16in. x 10in.
Layup: $[80/-65/80/-65]_s$

Optimization Problem Statement

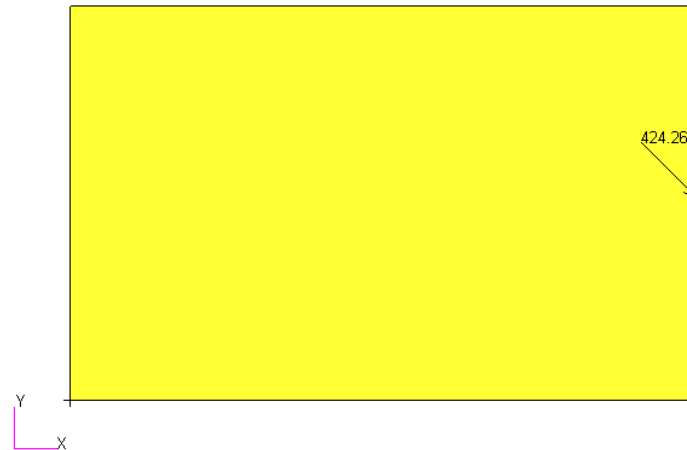
Design Region (Topometry)

z1: Thickness of ply 1 and 8
z2: Thickness of ply 2 and 7
z3: Thickness of ply 3 and 6
z4: Thickness of ply 4 and 5

$$.001 < z1, z2, z3, z4$$

Variable Linking

z3 = z1
z4 = z2



Design Objective

r0: Minimize compliance
This is equivalent to maximizing stiffness

Design Constraints

r1: Weight
 $r1 < 38.0$ (~50% of original weight 75.0)

Contact me

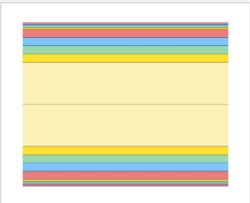
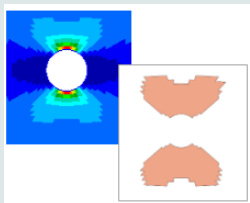
- Nastran SOL 200 training
- Nastran SOL 200 questions
- Structural or mechanical optimization questions
- Access to the SOL 200 Web App

christian@ the-engineering-lab.com

Before Continuing

Consider the New Composite Laminate Optimization Tutorials – Composite Coupon

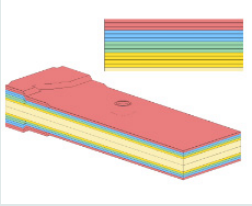
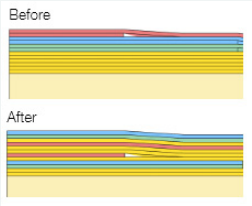
Visit the User's Guide to access the newest tutorials.

	Title and Description	PDF Tutorial	YouTube Tutorial
	<p>Composite Coupon – Phase A – Determination of the optimal 0° direction of a composite</p> <p>The goal of this 5-phase tutorial series is to optimize a composite coupon, with a core, and produce a lightweight composite that satisfies failure index constraints. The optimal ply shapes (ply drop-offs) and ply numbers are determined for 0°, ±45°, and 90° plies. A stacking sequence optimization is performed to satisfy manufacturing requirements. One important part of optimizing composites is visualizing the composite plies. This tutorial series also demonstrates the visualization of ply drop-offs, tapered plies and core layers.</p> <p>This first phase involves determining the optimal 0° direction of a composite. It is best practice to align the 0° plies in the direction of the load. Not doing so will more than likely produce a suboptimal composite that is heavier than necessary. This tutorial demonstrates the use of MSC Nastran's optimizer to determine the optimal 0° direction of a composite. An optimization is performed to maximize the stiffness of the composite for multiple load cases and while varying the angle of the 0° plies. Ultimately, the best 0° direction is determined.</p> <p>This is the first phase in a 5-phase tutorial series.</p>	Link	Link
	<p>Composite Coupon – Phase B – Baseline Ply Number Optimization</p> <p>This tutorial demonstrates how to configure a basic ply number optimization of continuous plies that span the entire model. The goal of this tutorial is to demonstrate basic actions such as creating variables, a weight objective and constraints on failure index. The results of this ply number optimization serve as a baseline for future comparisons. In a subsequent tutorial, the ply shapes will be optimized to minimize weight.</p> <p>This is the second phase in a 5-phase tutorial series.</p>	Link	Link
	<p>Composite Coupon – Phase C – Data Preparation for Ply Shape Optimization</p> <p>This tutorial is a guide to preparing data for ply shape optimization in a subsequent tutorial. The maximum failure index values of the outer plies of the composite are determined and saved to specially formatted PLY000i files. The PLY000i files will be used to construct optimal ply shapes in a subsequent tutorial.</p> <p>This is the third phase in a 5-phase tutorial series.</p>	Link	Link

Before Continuing

Consider the New Composite Laminate Optimization Tutorials – Composite Coupon, Continued

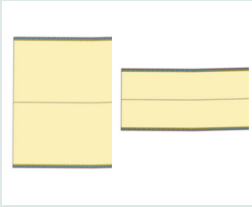

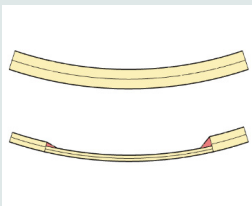
Visit the User's Guide to access the newest tutorials.

	Title and Description	PDF Tutorial	YouTube Tutorial
	Composite Coupon – Phase D – Ply Shape and Ply Number Optimization This tutorial details the process to build optimal ply shapes and perform a ply number optimization. The optimal ply shapes are constructed to follow the contours of the failure indices. The ply number optimization involves minimizing weight and constraining the failure indices of plies. The PLY000i files and BDF files from the previous tutorial, phase C, are used in this tutorial. This is the fourth phase in a 5-phase tutorial series.	Link	Link
	Composite Coupon – Phase E – Stacking Sequence Optimization This tutorial involves performing a stacking sequence optimization and is a continuation of the previous tutorial, phase D. A final statics analysis is performed to confirm the optimized composite satisfies failure index constraints. This is the fifth phase in a 5-phase tutorial series.	Link	Link

Before Continuing

Consider the New Composite Laminate Optimization Tutorials – Sandwich Composite Panel

Visit the User's Guide to access the newest tutorials.

	Title and Description	PDF Tutorial	YouTube Tutorial
	<p>Sandwich Composite Panel – Phase B – Baseline Core Thickness Optimization</p> <p>The goal of this 3-phase tutorial series is to optimize a curved composite panel, with a core, and produce a lightweight composite that satisfies constraints on the buckling load factor. This tutorial series focuses exclusively on optimizing the thickness of the core. The methods detailed in the tutorial series are applicable to both foam and honeycomb cores.</p> <p>This tutorial demonstrates how to configure a basic core thickness optimization where the core has a constant thickness throughout the entire model. The goal of this tutorial is to demonstrate basic actions such as creating variables, a weight objective and constraints on the buckling load factor. The results of this core thickness optimization serve as a baseline for future comparisons. In a subsequent tutorial, the core will be allowed to have a variable thickness throughout the model and will be optimized to minimize weight.</p> <p>This is the first phase in a 3-phase tutorial series.</p>	Link	Link
	<p>Sandwich Composite Panel – Phase C – Topometry Optimization to Determine Optimal Core Shape</p> <p>This tutorial is a guide to preparing data for core shape and core thickness optimization in a subsequent tutorial. A topometry optimization is performed in this tutorial to determine the ideal thickness distribution of the core throughout the entire composite panel while satisfying constraints on the buckling load factor and minimizing weight. The results of a topometry optimization are contained in the PLY000i files and will be used to construct optimal core shapes in a subsequent tutorial.</p> <p>This is the second phase in a 3-phase tutorial series.</p>	Link	Link
	<p>Sandwich Composite Panel – Phase D – Core Shape and Core Thickness Optimization</p> <p>This tutorial details the process to build optimal core shapes and perform a core thickness optimization. The optimal core shapes are constructed to follow the contours of thickness results generated by a topometry optimization. The core thickness optimization involves minimizing weight and constraining the buckling load factor. The PLY000i files and BDF files from the previous tutorial, phase C, are used in this tutorial. Comparisons are made between this optimization in phase D and the baseline optimization performed in phase B.</p> <p>This is the third phase in a 3-phase tutorial series.</p>	Link	Link

Tutorial

Tutorial Overview

1. Start with a .bdf or .dat file
2. Use the SOL 200 Web App to:
 - Convert the .bdf file to SOL 200
 - Design Regions/Variables
 - Design Objective
 - Design Constraints
 - Perform optimization with Nastran SOL 200
3. Review optimization results
 - .f06
 - Topometry Optimization and Structural Results

Special Topics Covered

Linked Properties in Topometry Optimization – When configuring a size optimization, there is an option to link variables, i.e. one variable is dependent on the change of another variable. A similar option exists for the thickness properties on the PCOMP entry, enabling users to link ply thicknesses of a composite laminate during a topometry optimization.

SOL 200 Web App Capabilities

The Post-processor Web App and HDF5 Explorer are free to MSC Nastran users.

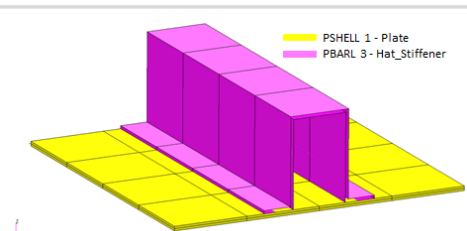
Compatibility

- Google Chrome, Mozilla Firefox or Microsoft Edge
- Windows and Red Hat Linux
- Installable on a company laptop, workstation or server. All data remains within your company.

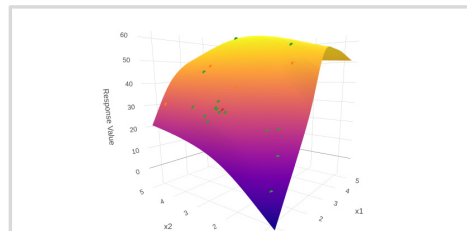
Web Apps

Benefits

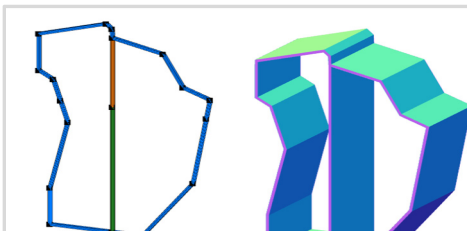
- REAL TIME error detection. 200+ error validations.
- REAL TIME creation of bulk data entries.
- Web browser accessible
- Free Post-processor web apps
- +80 tutorials



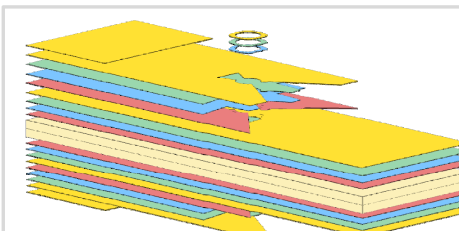
Web Apps for MSC Nastran SOL 200
Pre/post for MSC Nastran SOL 200.
Support for size, topology, topometry, topography, multi-model optimization.



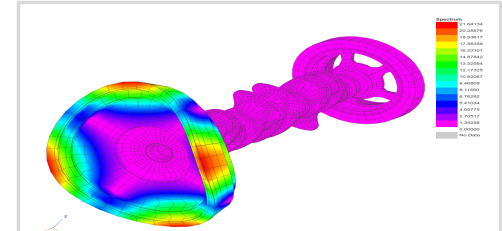
Machine Learning Web App
Bayesian Optimization for nonlinear response optimization (SOL 400)



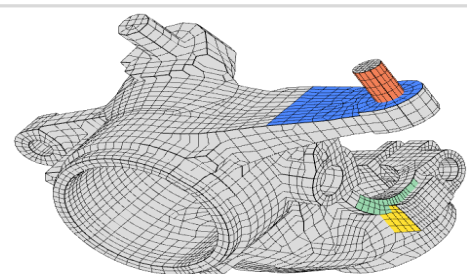
PBMSECT Web App
Generate PBMSECT and PBRSECT entries graphically



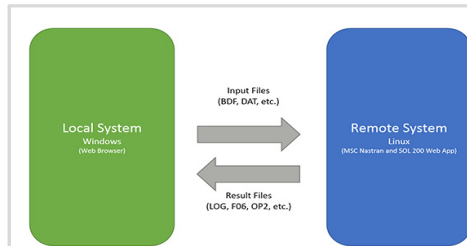
Ply Shape Optimization Web App
Optimize composite ply drop-off locations, and generate new PCOMPG entries



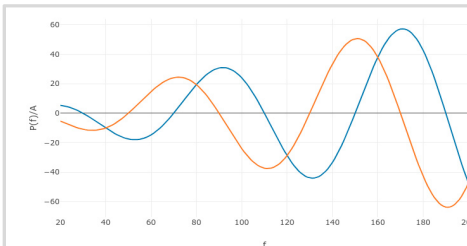
Post-processor Web App
View MSC Nastran results in a web browser on Windows and Linux



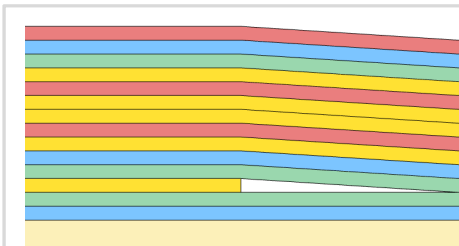
Shape Optimization Web App
Use a web application to configure and perform shape optimization.



Remote Execution Web App
Run MSC Nastran jobs on remote Linux or Windows systems available on the local network



Dynamic Loads Web App
Generate RLOAD1, RLOAD2 and DLOAD entries graphically



Stacking Sequence Web App
Optimize the stacking sequence of composite laminate plies

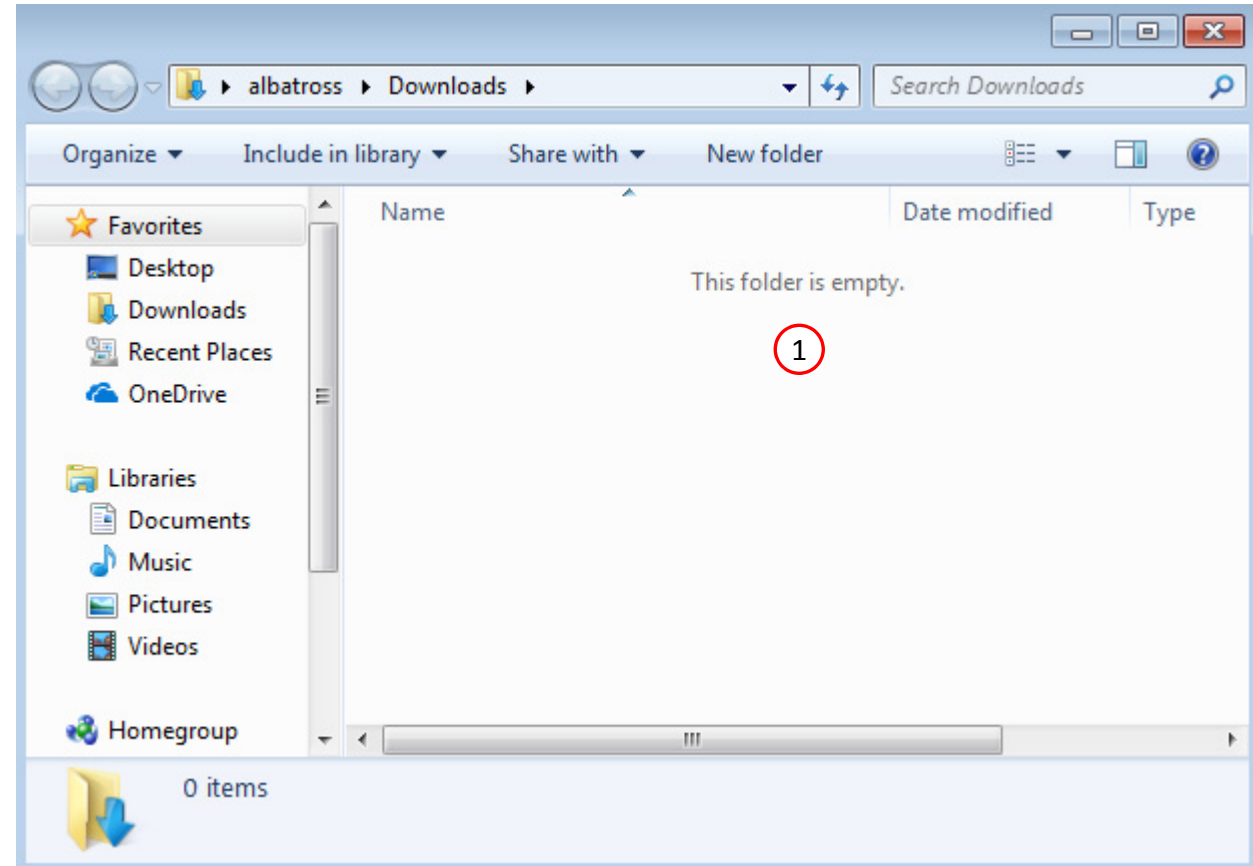


HDF5 Explorer Web App
Create graphs (XY plots) using data from the H5 file

Before Starting

1. Ensure the Downloads directory is empty in order to prevent confusion with other files

- Throughout this workshop, you will be working with multiple file types and directories such as:
 - .bdf/.dat
 - nastran_working_directory
 - .f06, .log, .pch, .h5, etc.
- To minimize confusion with files and folders, it is encouraged to start with a clean directory.



Go to the User's Guide

1. Click on the indicated link

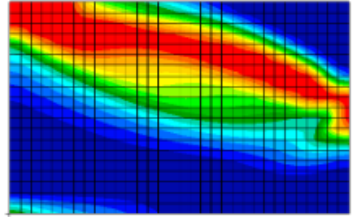
- The necessary BDF files for this tutorial are available in the Tutorials section of the User's Guide.



Obtain Starting Files

1. Find the indicated example
2. Click Link
3. The starting file has been downloaded

- When starting the procedure, all the necessary BDF files must be collected together.



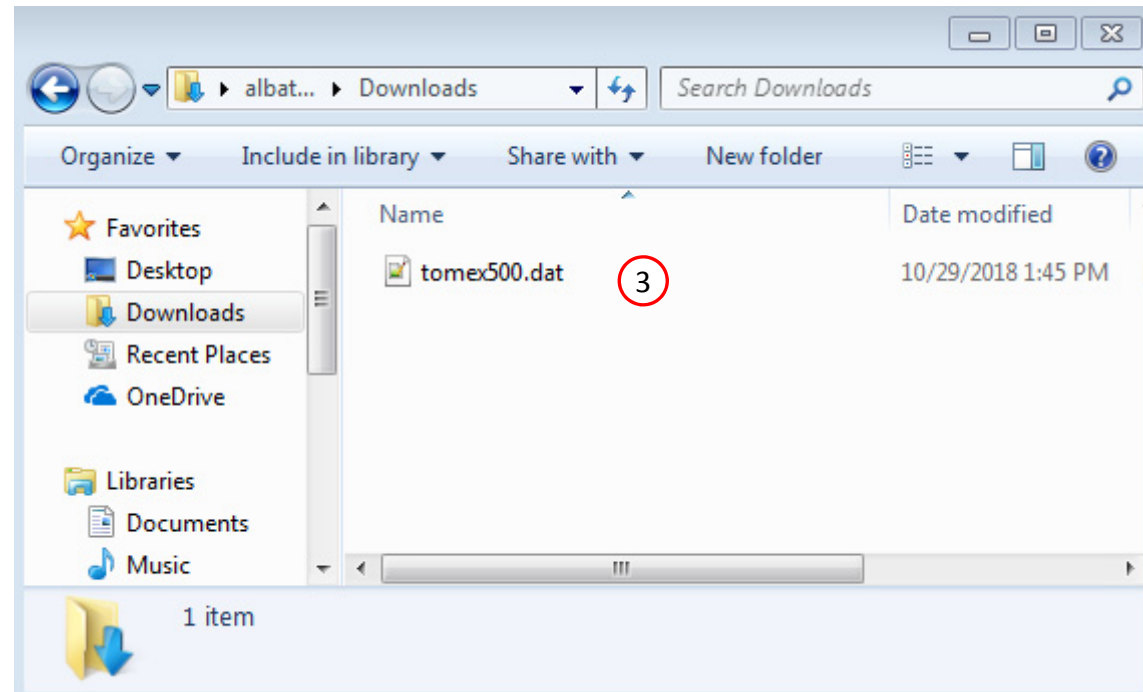
1

MSC Nastran Topometry Optimization of a Composite Panel

This tutorial covers the use of Topometry Optimization to determine ply shapes.

Starting BDF Files: [Link](#) **2**

Solution BDF Files: [Link](#)



Open the Correct Page

1. Click on the indicated link

- MSC Nastran can perform many optimization types. The SOL 200 Web App includes dedicated web apps for the following:
 - Optimization for SOL 200 (Size, Topology, Topometry, Topography, Local Optimization, Sensitivity Analysis and Global Optimization)
 - Multi Model Optimization
 - Machine Learning
- The web app also features the HDF5 Explorer, a web application to extract results from the H5 file type.



Upload BDF Files

1. Click 1. Select Files and select tomex500.dat
2. Click Upload Files

- The process starts by uploading all the necessary BDF files. The BDF files can be files of your own or files found in the Tutorials section of the User's Guide.

Step 1 - Upload .BDF Files

The screenshot shows a two-step process for uploading files. Step 1, '1. Select files', is highlighted with a blue button and shows the file 'tomex500.dat' selected. Below it is a green progress bar labeled 'Inspecting: 100%'. Step 2, '2. Upload files', is highlighted with a green button. Below it is another green progress bar labeled 'Uploading: 100 %'. At the bottom, there is a checkbox labeled 'List of Selected Files' which is currently unchecked.

1. Select files tomex500.dat

Inspecting: 100%

2. Upload files

Uploading: 100 %

☐ List of Selected Files

Create Design Region

1. Click Topometry
2. In the search box, type 'thickness'
3. Click on the plus (+) icons to set the thickness (Ti) of Plies 1-4 as Design Regions
4. Scroll to the section titled Step 2 – Adjust TOMVAR entries
5. The new Design Region is added to the table, no further edit is necessary

- Suppose the goal is to vary the thickness. In traditional Size optimization, the thickness can be a set a single design variable. With Topometry optimization, when the design region is set, each element in the region is given its own independent thickness design variable.
- If PSHELL 1 has 500 elements associated and is configured as a design region, then there will be 500 design variables created.
- Each step has hidden functionality for advanced users. The visibility is controlled by clicking +Options .
- If the property entry, e.g. PSHELL, was given a name in Patran, e.g. Car Door, the name can be shown by marking the checkbox titled Entry Name.

Size Topology **Topometry** **1** ography

Step 1 - Select design properties

+ Options

Create TOMVAR	Property ⇅	Property Description ⇅	Entry ⇅	Entry ID ⇅	Current Value ⇅
	<input type="text" value="Search"/>	<input <b="" type="text" value="thickness"/> 2	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>
3 <input type="checkbox"/>	T1	Thickness of ply	PCOMP	1	1.000
<input type="checkbox"/>	T2	Thickness of ply	PCOMP	1	1.000
<input type="checkbox"/>	T3	Thickness of ply	PCOMP	1	1.000
<input type="checkbox"/>	T4	Thickness of ply	PCOMP	1	1.000

5 10 20 30 40 50
Number of Visible Rows 5

4 Step 2 - Adjust TOMVAR Entries

+ Options

✕ Delete Visible Rows





	Label ⇅	Status ⇅	Property ⇅	Property Description ⇅	Entry ⇅	Entry ID ⇅	Initial Value ⇅	Lower Bound	Upper Bound	Allowed Discrete Values
	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>
<input checked="" type="checkbox"/>	z1	<input checked="" type="checkbox"/>	T1	Thickness of ply	PCOMP	1	1.000	.001	Upper	Examples: -2.0, 1.0, THRU, 10.0, B'
<input checked="" type="checkbox"/>	z2	<input checked="" type="checkbox"/>	T2	Thickness of ply	PCOMP	1	1.000	.001	Upper	Examples: -2.0, 1.0, THRU, 10.0, B'
<input checked="" type="checkbox"/>	z3	<input checked="" type="checkbox"/>	T3	Thickness of ply	PCOMP	1	1.000	.001	Upper	Examples: -2.0, 1.0, THRU, 10.0, B'
<input checked="" type="checkbox"/>	z4	<input checked="" type="checkbox"/>	T4	Thickness of ply	PCOMP	1	1.000	.001	Upper	Examples: -2.0, 1.0, THRU, 10.0, B'

Create Design Region

1. Scroll to the section titled Step 3 – Optional – Create variable links for TOMVARs
2. Click 2 times on +Create DLINK
3. Create design variables links for the thickness of ply 1 and 3
 1. Dependent Design Variables: z3
 2. Equation: z1
4. Create design variables links for the thickness of ply 2 and 4
 1. Dependent Design Variables: z4
 2. Equation: z2

- It is important to verify the Equation is configured properly. For example, the variable z3 is initially equal to 1.0. When the Equation is configured, it should also produce an initial value of 1.0. The resulting value of the Equation is displayed on the column titled Value of Equation and can be used to validate the Equation is configured properly.

Step 3 - Create variable links for TOMVARs

					2 + Create DLINK
	Status ▾	Dependent Design Variables ▾	Equation (Independent Design Variables) ▾	Value of Equation ▾	
	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	
		z3 3.1	z1 3.2	1.	
		z4 4.1	z2 4.2	1.	

Create Design Objective

1. Click on Objective
2. Type 'comp' in the search box
3. Select the plus(+) icon for Compliance
4. The objective with label r0 is created.
The objective is to minimize (MIN)

• The objective must always be a single scalar response. A response such as weight and volume are single responses and can be used as an objective. Other responses require special care when set as an objective. For example, if the objective is stress, only the stress of a single component, e.g. von Mises, of a single element, of a single load case may be used.

Objective

Equation Objective

1

Step 1 - Select an objective

Select an analysis type

SOL 101 - Statics

Select a response

	Response Description ▾	Response Type ▾
	<input type="text" value="Search"/>	<input type="text" value="comp"/>
3 	Compliance (Product of displacement and the applied load)	COMP

5

10

20



30

40

50

Step 2 - Adjust objective

+ Options

	Label	Status	Response Type	Maximize or Minimize	Property Type	ATTA	ATTB	ATTI
	r0		COMP	4 MIN ▾				

Create Design Constraints

1. Click Constraints
2. Click on the plus (+) icon for Weight
3. Configure the following for r1:
 - Upper Allowed Limit: 38.0

- The constraints are defined normally as would be done in a size optimization.

Step 1 - Select constraints

Select an analysis type

SOL 101 - Statics

Select a response

	Response Description ▾	Response Type ▾
	<input type="text" value="Search"/>	<input type="text" value="Search"/>
2 +	Weight	WEIGHT
+	Volume	VOLUME
+	Displacement	DISP
+	Strain	STRAIN
+	Element Strain Energy	ESE

«
1
2
3
4
5
»

5
10
20
30
40
50

Step 2 - Adjust constraints

+ Options

	Label ▾	Status ▾	Response Type ▾	Property Type ▾	ATTA ▾	ATTB ▾	ATTi ▾	Lower Allowed Limit	Upper Allowed Limit
	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>	<input type="text" value="Search"/>
x	r1	✓	WEIGHT		3 ▾	3 ▾		Lower	38.0

Configure Optimization Settings

1. Click Settings
2. Set P2 to 12 – Print constraints and responses

- The P2 setting controls the output of the following information to the F06 file: objective, constraints, responses, properties and design variables.
- This is a topometry optimization and will generate a large amount of property and design variable data in the F06 file. To make the F06 file size manageable, the design variable information is omitted by using the P2=12 option. When the results are viewed, note that the objective and constraint information is plotted, but the design variable history is not plotted due to the P2=12 option.
- If this is a combined size and topometry optimization, P2 should be set to 15. If this is a pure size optimization, P2 should be set to 15.

1

Optimization Settings

Parameter ▾	Description ▾	Configure ▾
Search	Search	Search
APRCOD	Approximation method to be used	<input type="checkbox"/> 2 - Mixed Method ▾
CONV1	Relative criterion to detect convergence	<input type="checkbox"/> Enter a positive real number
CONV2	Absolute criterion to detect convergence	<input type="checkbox"/> Enter a positive real number
DELX	Fractional change allowed in each design variable during any optimization cycle	<input type="checkbox"/> Enter a positive real number
DESMAX	Maximum number of design cycles to be performed	<input checked="" type="checkbox"/> 20
DISBEG	Design cycle number for discrete variable processing initiation	<input type="checkbox"/> Enter a positive integer
GMAX	Maximum constraint violation allowed at the converged optimum	<input type="checkbox"/> Enter a positive real number
P1	Print items, e.g. objective, design variables, at every n-th design cycle to the .f06 file	<input checked="" type="checkbox"/> 1
P2	Items to be printed to the .f06 file	<input checked="" type="checkbox"/> 12 - Print constraints and responses ▾
TCHECK	Topology Checkerboarding	<input type="checkbox"/> -1 - Automatic selection (Default) ▾
TDMIN	Minimum diameter of members in topology optimization	<input type="checkbox"/> Enter a positive real number
TREGION	Trust Region	<input type="checkbox"/> 1 - Trust Region On ▾

2

Export New BDF Files

1. Click on Exporter
2. Click on Download BDF Files

- When the download button is clicked a new file named "nastran_working_directory" is downloaded. If the file already exists in your local folder, the folder name is appended with a number, e.g. "nastran_working_directory (1).zip"

SOL 200 Web App - Optimization Upload Variables Objective Constraints Subcases **1** Exporter Results Settings Match Other User's Guide Home

BDF Output - Model

```
assign userfile = 'optimization_results.csv', status = unknown,
form = Formatted, unit = S2
$ id msc, tomex.dat $ xmy 3-Mar-2009 mdr4
$ PCOMP PLY-BY-PLY TOPOMETRY OPT
SOL 200
CEND

SEALL = ALL
SUPER = ALL
TITLE = PCOMP PLY-BY-PLY TOPOMETRY WITH IPOPT
ECHO = NONE
MAXLINES = 999999999
  DESOBJ(MIN) = 8000000
  DESGLB = 40000000
  $ DSAPRT(FORMATTED, EXPORT, END=SENS) = ALL
SUBCASE 1
  ANALYSIS = STATICS
  $ DESOBJ slot
  $ DRSPAN slot
  SUBTITLE=Default
  SPC = 1
  LOAD = 3
  displacement(plot,sort1,real)=all
  spcforces(plot,sort1,real)=all
  stress(plot,sort1,real,vonmises,center)=all
BEGIN BULK
INCLUDE './design_model.bdf'
```

Download BDF Files

[Download BDF Files](#) **2**

BDF Output - Design Model

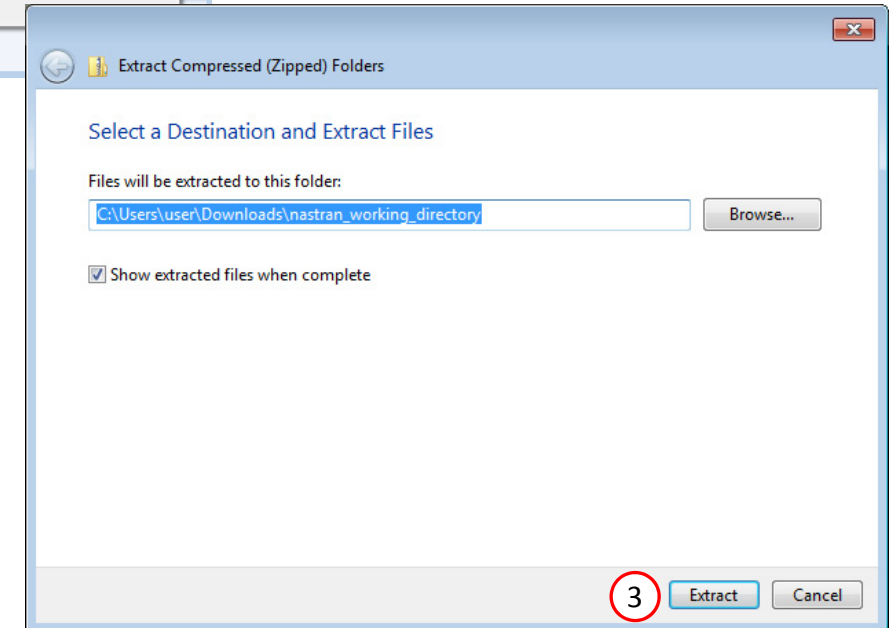
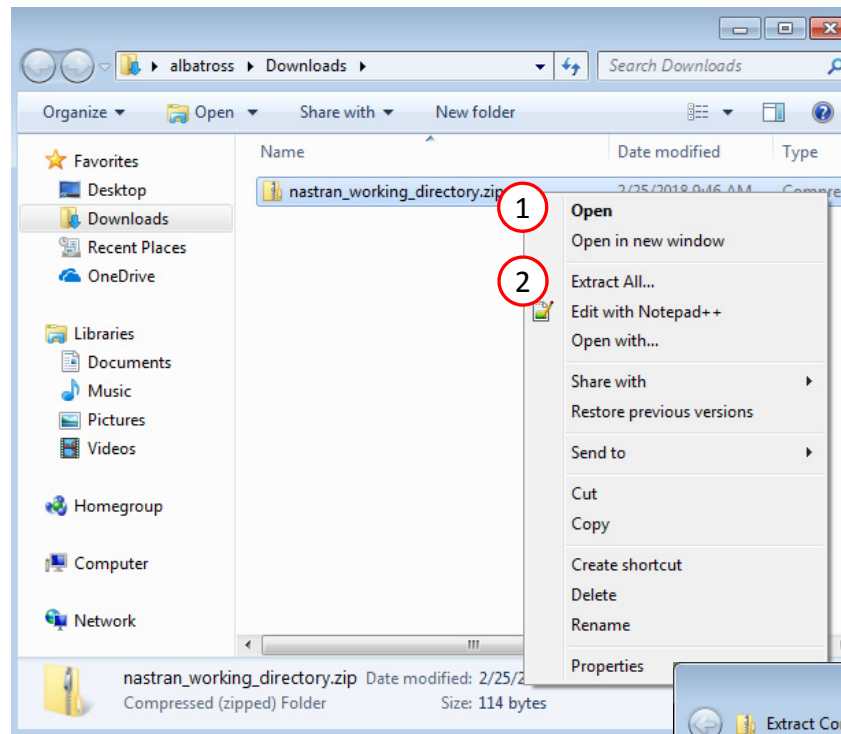
```
$*****
$
$          Design Model
$
$*****
$
$          Design Regions/Variables
$-----
$
$
TOMVAR 3000001 PCOMP 1 T1 1.000 .001
TOMVAR 3000002 PCOMP 1 T2 1.000 .001
TOMVAR 3000003 PCOMP 1 T3 1.000 .001
      DLINK 3000001 1.0
TOMVAR 3000004 PCOMP 1 T4 1.000 .001
      DLINK 3000002 1.0
$
$
$
$          Design Objective
$-----
$
$
DRESP1 8000000 r0 COMP
$
$
$          Design Constraints
$-----
$
$
DRESP1 8000001 r1 WEIGHT 3 3
$
$
$
DCONSTR 30001 8000001 30.
$
$          Design Equation Constraints
$-----
$
$
$
$
```

Developed by The Engineering Lab

Perform the Optimization with Nastran SOL 200

1. A new .zip file has been downloaded
2. Right click on the file
3. Click Extract All
4. Click Extract on the following window

- Always extract the contents of the ZIP file to a new, empty folder.



Perform the Optimization with Nastran SOL 200

1. Inside of the new folder, double click on Start MSC Nastran
2. Click Open, Run or Allow Access on any subsequent windows
3. MSC Nastran will now start

- After a successful optimization, the results will be automatically displayed as long as the following files are present: BDF, F06 and LOG.
- One can run the Nastran job on a remote machine as follows:
 - 1) Copy the BDF files and the INCLUDE files to a remote machine.
 - 2) Run the MSC Nastran job on the remote machine.
 - 3) After completion, copy the BDF, F06, LOG, H5 files to the local machine.
 - 4) Click "Start MSC Nastran" to display the results.

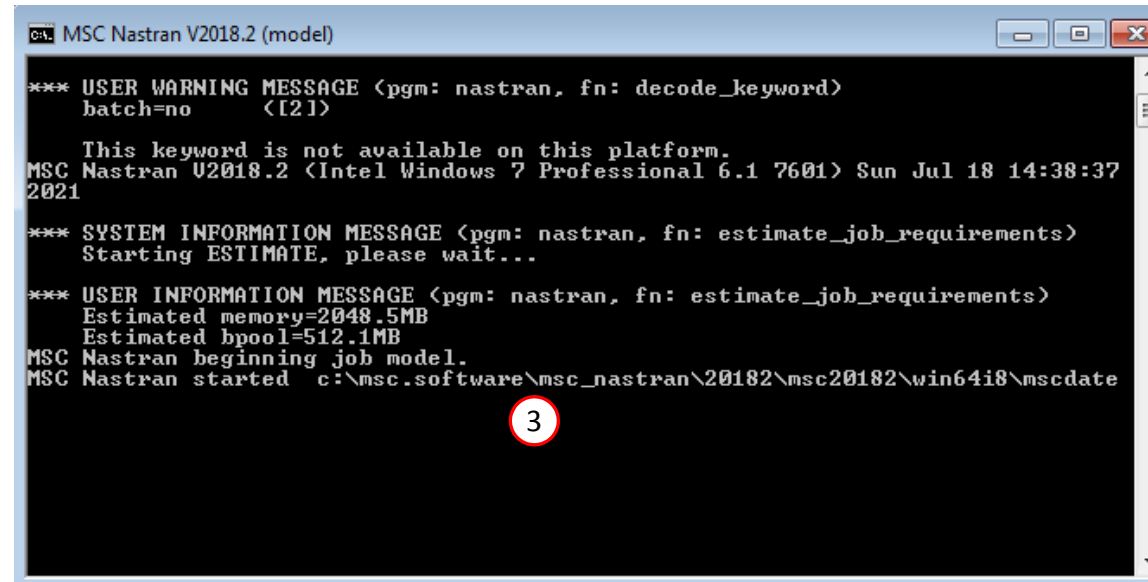
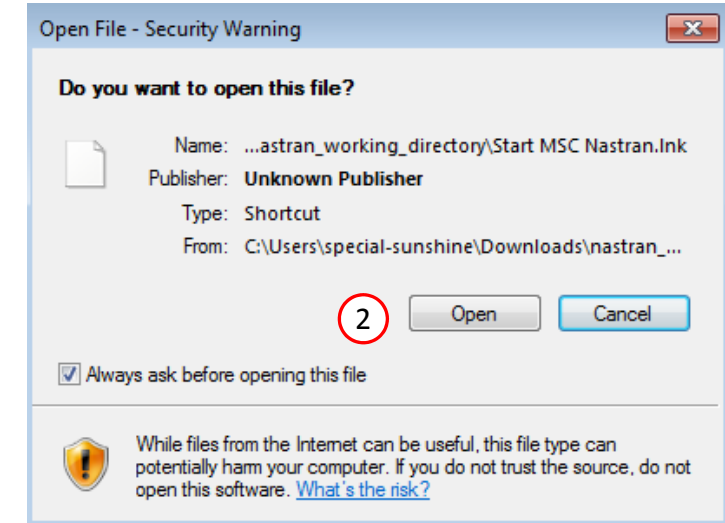
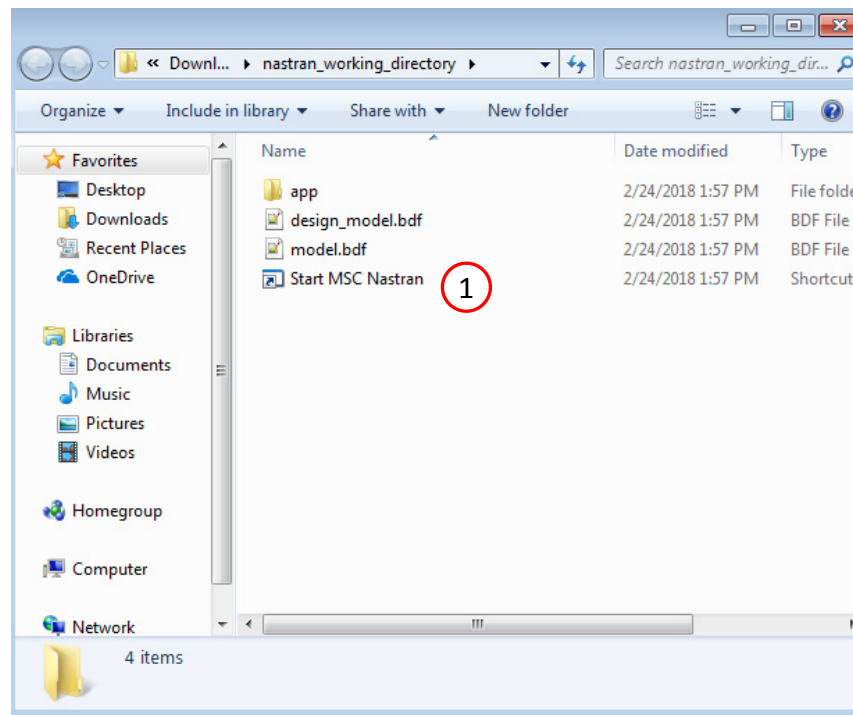
Using Linux?

Follow these instructions:

- 1) Open Terminal
- 2) Navigate to the nastran_working_directory
`cd ./nastran_working_directory`
- 3) Use this command to start the process
`./Start_MSC_Nastran.sh`

In some instances, execute permission must be granted to the directory. Use this command. This command assumes you are one folder level up.

```
sudo chmod -R u+x ./nastran_working_directory
```



Status

1. While MSC Nastran is running, a status page will show the current state of MSC Nastran

- The status of the MSC Nastran job is reported on the Status page. Note that Windows 7 users will experience a delay in the status updates. All other users of Windows 10 and Red Hat Linux will see immediate status updates.

SOL 200 Web App - Status

 Python

 MSC Nastran

Status

Name	Status of Job	Design Cycle	RUN TERMINATED DUE TO
model.bdf	Running	None	

Review Optimization Results

After MSC Nastran is finished, the results will be automatically uploaded.

1. Ensure the messages shown have green checkmarks. This is indication of success. Any red icons indicate challenges.
2. The final value of objective, normalized constraints (not shown here) and design variables can be reviewed.

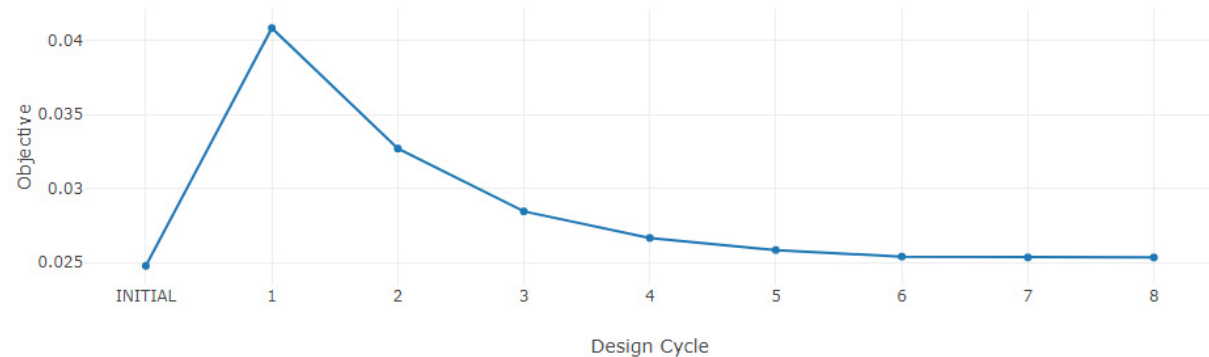
- Note that in a Topometry optimization, hundreds or thousands of design variables can be created. In this situation, the Design Variables are not plotted and displayed. Instead, the Objective and Normalized Constraints are displayed. It is recommended that a traditional post-processor be used to review the design variable results.

1 Final Message in .f06



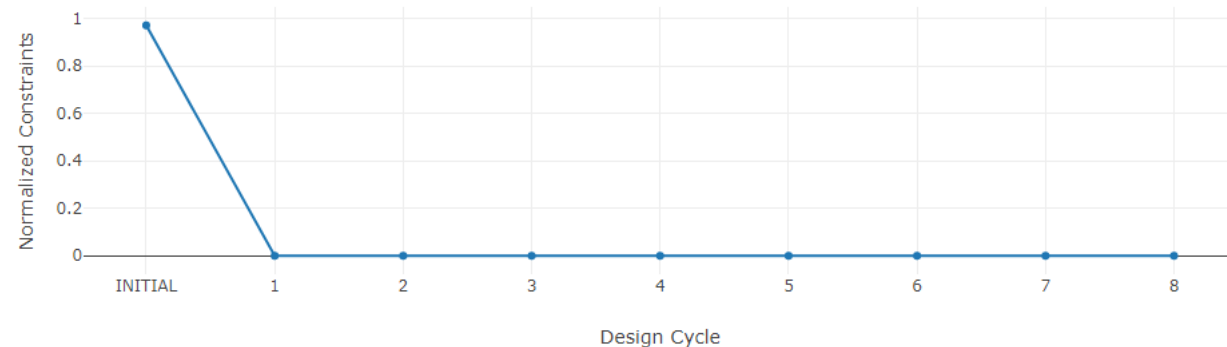
RUN TERMINATED DUE TO HARD CONVERGENCE TO AN OPTIMUM AT CYCLE NUMBER = 8.

Objective



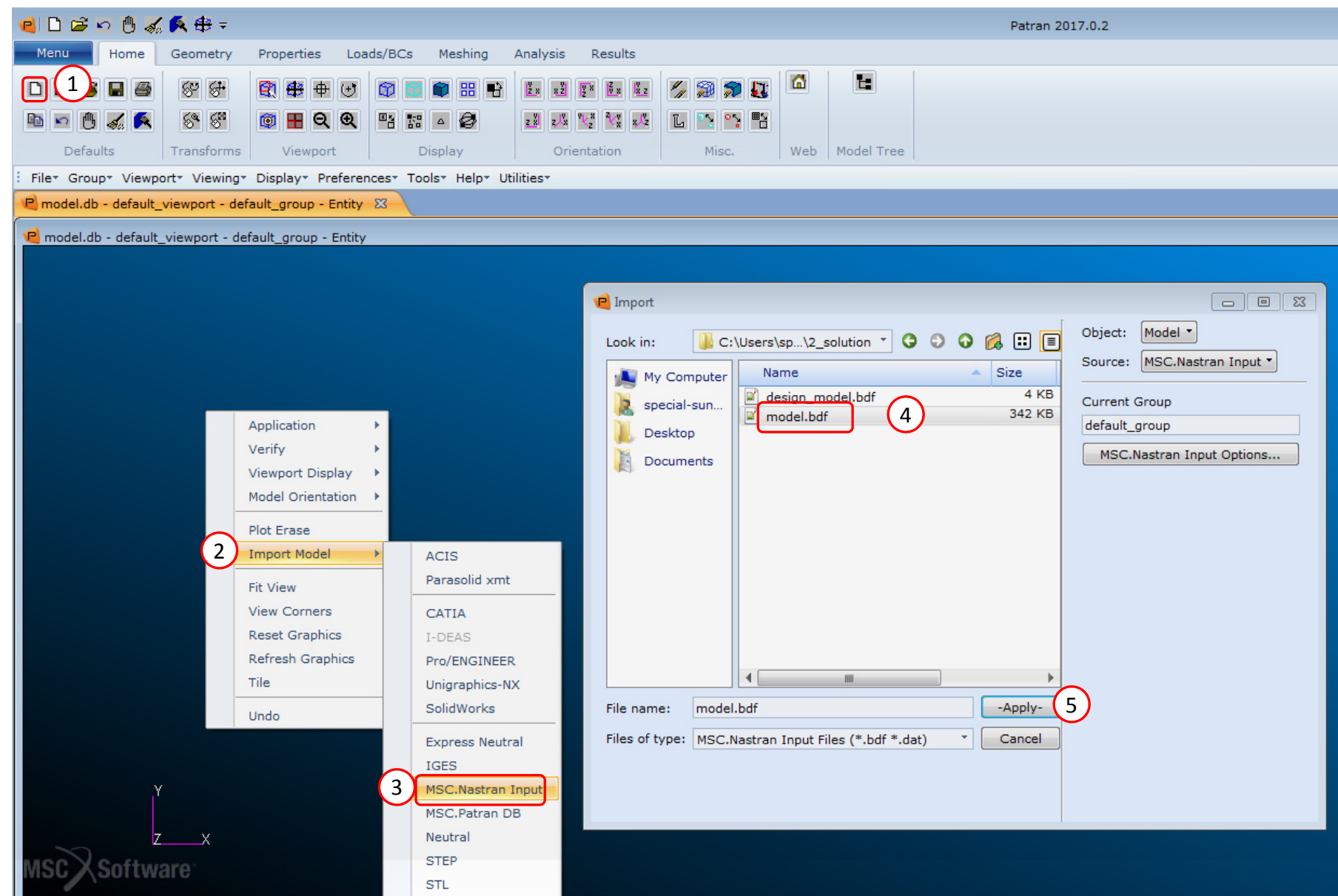
2 Normalized Constraints

+ Info



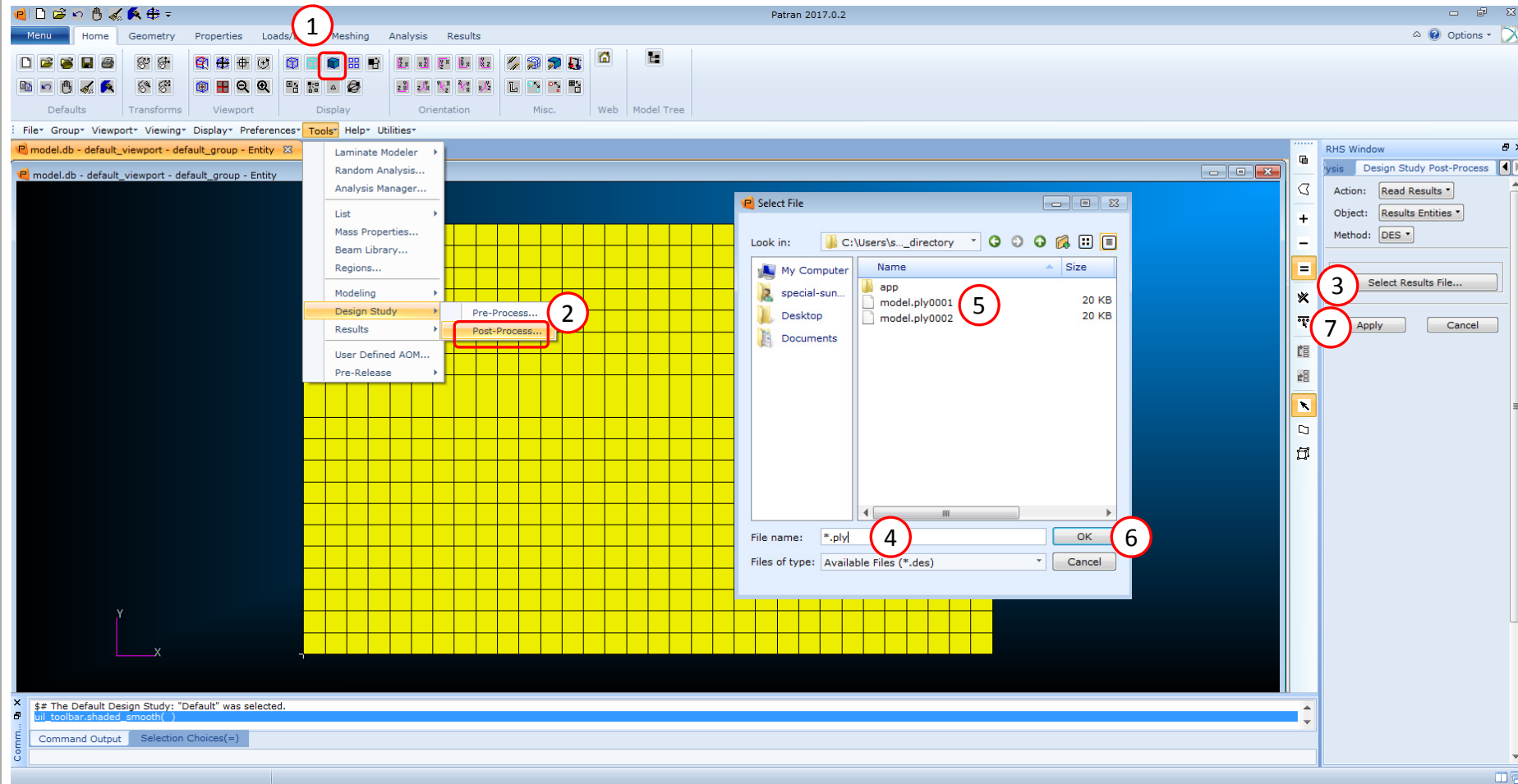
Review Optimization Results in Patran

1. Start a new Patran session
2. Right click to open a menu
3. Go to Import Model and click on MSC.Nastran Input
4. Select model.bdf (This file was used for the optimization)
5. Click Apply



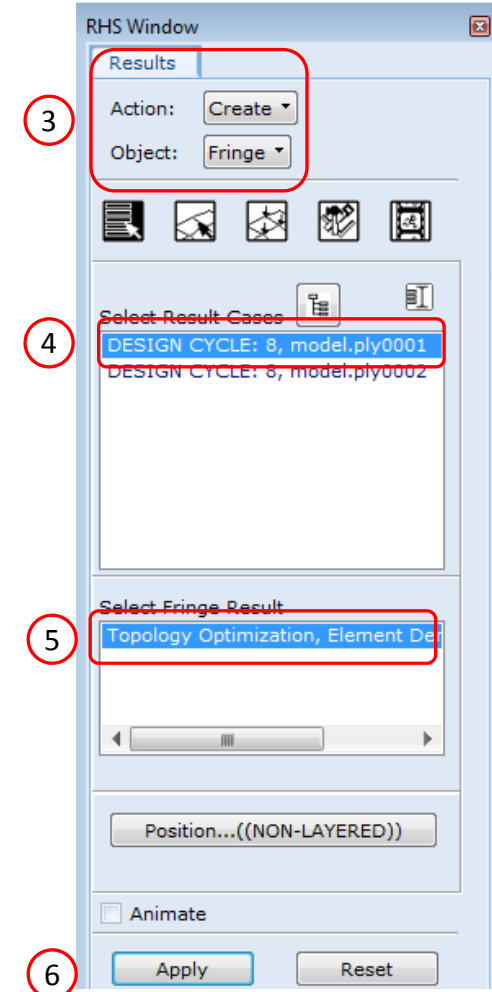
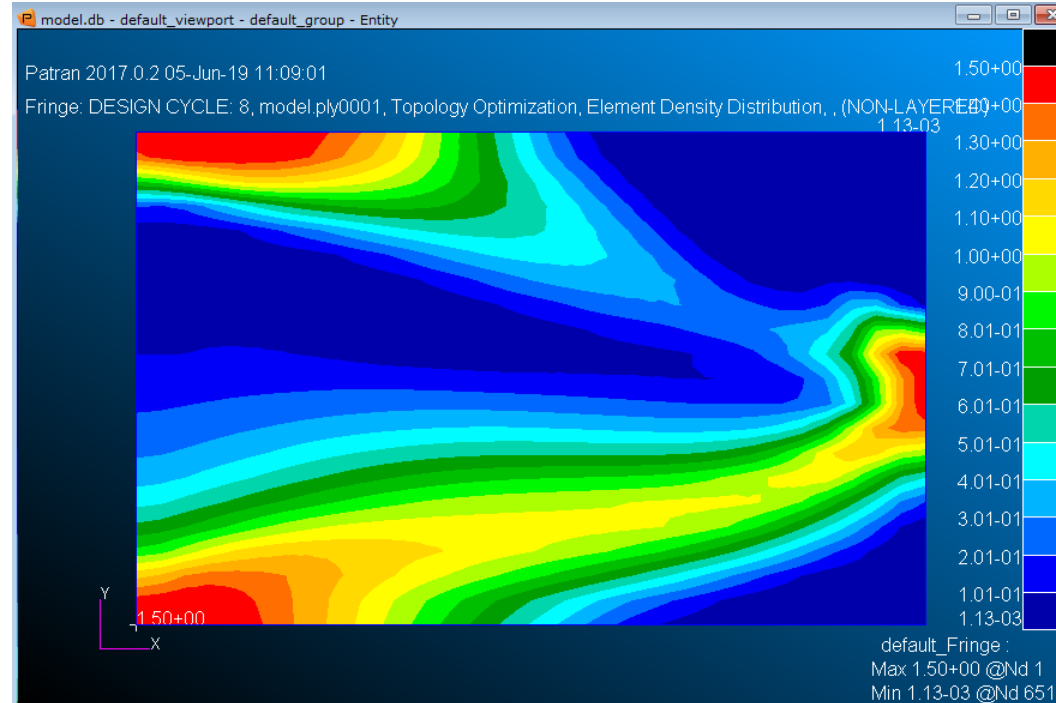
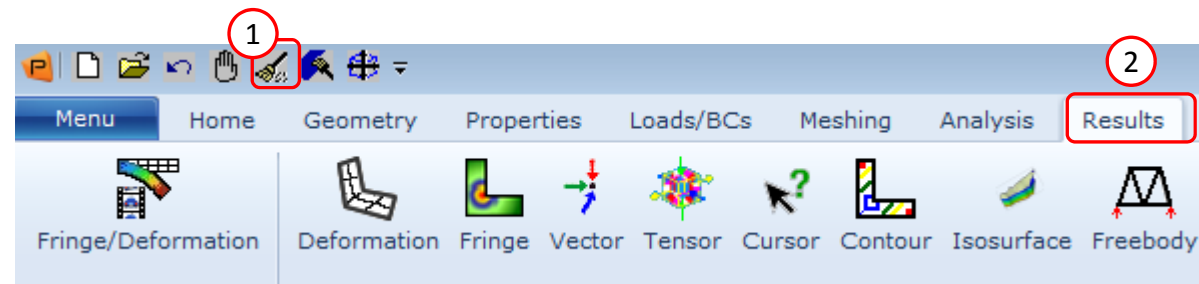
Review Optimization Results in Patran

1. Click Smooth Shading
2. Go to Tools > Design Study and click on Post-Process
3. Click Select Results File
4. Type "*.ply" and press the enter key to show all the files in the folder
5. Select model.ply0001 (This file was created during the optimization)
6. Click OK
7. Click Apply
8. Repeat steps 3-7 for the following files:
 1. model.ply0002



Review Optimization Results in Patran

1. Click the clear icon
2. Click Results
3. Set the following:
 - Action: Create
 - Object: Fringe
4. Select: DESIGN CYCLE: 8, model.ply0001
(The row should be highlighted blue)
5. Select Topology Optimization, Element D...
(The row should be highlighted blue)
6. Click Apply
7. Repeat steps 4-6 for the following:
 - DESIGN CYCLE: 8, model.ply0002



Update the Original Model

Ensure the BDF files prior to optimization have one of these entries:

- H5 Output
 - MDLPRM HDF5 1
 - HDF5OUT INPUT YES

MDLPRM HDF5 is supported in MSC Nastran 2016.1 and newer. HDF5OUT is supported in MSC Nastran 2022.2 and newer.

The following applies to MSC Nastran 2023.4 and older. For MSC Nastran 2024.1, this is not needed.

Change DESPCH1 to 1

- DESPCH
 - Before:
PARAM DESPCH1 -1
 - After:
PARAM DESPCH1 1

DESPCH1 -1 outputs entries to the PCH file in the small field format. Since the PSHELL IDs are longer than 8 characters, the IDs appear as asterisk characters, e.g. *****. DESPCH1 1 outputs the entries in the large field format, so the IDs are fully visible.

tomex500.dat

24	CQUAD4	1	1	2	23	22
25	CQUAD4	2	1	3	24	23
26	CQUAD4	3	1	4	25	24
27	CQUAD4	4	1	5	26	25
28	CQUAD4	5	1	6	27	26
29	CQUAD4	6	1	7	28	27
30	CQUAD4	7	1	8	29	28
31	CQUAD4	8	1	9	30	29
32	CQUAD4	9	1	10	31	30
33	CQUAD4	10	1	11	32	31
34	CQUAD4	11	1	12	33	32
35	CQUAD4	12	1	13	34	33
36	CQUAD4	13	1	14	35	34
37	CQUAD4	14	1	15	36	35
38	CQUAD4	15	1	16	37	36
39	CQUAD4	16	1	17	38	37
40	CQUAD4	17	1	18	39	38

2

tomex500 (1).dat

24	CQUAD4	1	10000001	1	2	23	22
25	CQUAD4	2	10000002	2	3	24	23
26	CQUAD4	3	10000003	3	4	25	24
27	CQUAD4	4	10000004	4	5	26	25
28	CQUAD4	5	10000005	5	6	27	26
29	CQUAD4	6	10000006	6	7	28	27
30	CQUAD4	7	10000007	7	8	29	28
31	CQUAD4	8	10000008	8	9	30	29
32	CQUAD4	9	10000009	9	10	31	30
33	CQUAD4	10	10000010	10	11	32	31
34	CQUAD4	11	10000011	11	12	33	32
35	CQUAD4	12	10000012	12	13	34	33
36	CQUAD4	13	10000013	13	14	35	34
37	CQUAD4	14	10000014	14	15	36	35
38	CQUAD4	15	10000015	15	16	37	36
39	CQUAD4	16	10000016	16	17	38	37
40	CQUAD4	17	10000017	17	18	39	38

```

1 $
2 $
3 $
4 $ CONTINUOUS DESIGN CYCLE NUMBER = 8 $
5 $
6 $
7 $
8 $
9 $
10 $
11 $ CONTINUOUS DESIGN CYCLE NUMBER = 8 $
12 $
13 $
14 $
15 $
16 $ UPDATED ANALYSIS MODEL DATA ENTRIES
17 $
18 PCOMP 10000001-.0105 0.0 650000. TSAI 0.0 0.0 SYM
19 70 1.5 80. YES 70 .774108-65. YES
20 70 1.5 80. YES 70 .774108-65. YES
21 $ Spawned PSHELL, MAT2 entries from PCOMP 10000001
22 $ PSHELL* 10000001 110000001 9.09643308E+00 210000001
23 $ * 1.000000000E+00 0 1.000000000E+00 0.000000000E+00
24 $ * -1.05000000E-02 9.08593308E+00 410000001
25 $ MAT2* 110000001 1.92043294E+06 2.00203393E+06 -4.72326628E+05
26 $ * 2.27776275E+07 3.50255102E+04 2.62296904E+06 5.85260000E-02
27 $ * 0.00000000E+00 0.00000000E+00 0.00000000E+00 0.00000000E+00
28 $ * 0.00000000E+00 0.00000000E+00 0.00000000E+00 0.00000000E+00
29 $ * 0

```

PCH

	1	2	3	4	5	6	7	8	9
1384	\$ Ultimate longitudinal compressive strength: 217.56E3 PSI								
1385	\$ Ultimate transverse tensile strength: 5.802E3 PSI								
1386	\$ Ultimate transverse compressive strength: 35.68E3 PSI								
1387	\$ Ultimate in-plane shear strength: 9.863E3 PSI								
1388	\$								
1389	\$ Source: Tsai, S.W. and Hahn, H.T., Introduction to Composite Materials								
1390	\$ FL, Table 1.7, p. 19; Table 7.1, p.292; Table 8.3, p. 342.								
1391	\$ 1	2	3	4	5	6	7	8	9
1392	MAT8	70	26.25e6	1.49e6	.28	1.04e6			.05
1393					217560.	217560.	5802.	35683.	986
1394	PCOMP	10000001-.0105	0.0		650000.	TSAI	0.0	0.0	
1395		70 1.5	80.		YES	70	.774108-	65.	
1396		70 1.5	80.		YES	70	.774108-	65.	
1397	PCOMP	10000002-.0105	0.0		650000.	TSAI	0.0	0.0	
1398		70 1.39312	80.		YES	70	.052964-	65.	
1399		70 1.39312	80.		YES	70	.052964-	65.	
1400	PCOMP	10000003-.0105	0.0		650000.	TSAI	0.0	0.0	
1401		70 .946061	80.		YES	70	.011974-	65.	
1402		70 .946061	80.		YES	70	.011974-	65.	
1403	PCOMP	10000004-.0105	0.0		650000.	TSAI	0.0	0.0	
1404		70 .713549	80.		YES	70	.001135-	65.	
1405		70 .713549	80.		YES	70	.001135-	65.	
1406	PCOMP	10000005-.0105	0.0		650000.	TSAI	0.0	0.0	
1407		70 .534361	80.		YES	70	.001135-	65.	
1408		70 .534361	80.		YES	70	.001135-	65.	
1409	PCOMP	10000006-.0105	0.0		650000.	TSAI	0.0	0.0	
1410		70 .399395	80.		YES	70	.001142-	65.	
1411		70 .399395	80.		YES	70	.001142-	65.	
1412	PCOMP	10000007-.0105	0.0		650000.	TSAI	0.0	0.0	
1413		70 .317016	80.		YES	70	.00312	-65.	
1414		70 .317016	80.		YES	70	.00312	-65.	

Downloaded BDF/DAT File

Update the Original Model

1. This Python script is used to automate the update the process.

```
import h5py
import hdf5plugin # This library is necessary when HDF5OUT is used (Approximately MSC Nastran 2021 and newer)
import re

def get_dataset_cquad4(path_of_h5_file):
    file = h5py.File(path_of_h5_file, 'r')
    dataset1 = file['/NASTRAN/INPUT/ELEMENT/CQUAD4']
    dataset_original = dataset1[...]

    list_of_objects = []

    for element in dataset_original:
        # Store the following fields EID, PID, G1, G2, G3, G4
        list_of_objects.append(
            {
                'eid': element[0],
                'pid': element[1],
                'g1': element[2][0],
                'g2': element[2][1],
                'g3': element[2][2],
                'g4': element[2][3]
            }
        )

    return list_of_objects

def read_cquad4_entries_from_h5_and_write_to_bdf(path_a, path_of_new_bdf_file):
    objects_a = get_dataset_cquad4(path_a)
    list_of_strings = []

    for element_i in objects_a:
        # Write the fields to an array/list
        # Ensure all array elements are strings so ','.join() works properly
        array_of_fields = [
            'CQUAD4',
            str(element_i['eid']),
            str(element_i['pid']),
            str(element_i['g1']),
            str(element_i['g2']),
            str(element_i['g3']),
            str(element_i['g4'])
        ]

        # Create the entry with comma delimiters, which is the free field format
        list_of_strings.append(','.join(array_of_fields))

    # Write the strings to a text file
    file = open(path_of_new_bdf_file, 'w')

    for item in list_of_strings:
        file.write(item + '\n')

    file.close()

def filter_entries_from_pch(path_of_pch_file, name_of_entry, path_of_new_bdf_file):
    # This function reads a PCH file and keeps specific entries
    # Before (PCH File):
    # PCOMP 10000001-.0105 0.0 650000. TSAI 0.0 0.0 SYM
    # 70 1.5 80. YES 70 .774108-65. YES
    # 70 1.5 80. YES 70 .774108-65. YES
    # $ Spawned PSHELL, MAT2 entries from PCOMP 10000001
    # $ PSHELL* 10000001 110000001 9.09643308E+00 210000001*
    # $ * 1.00000000E+00 0 1.00000000E+00 0.00000000E+00*
    # After (new entries.bdf):
    # PCOMP 10000001-.0105 0.0 650000. TSAI 0.0 0.0 SYM
    # 70 1.5 80. YES 70 .774108-65. YES
    # 70 1.5 80. YES 70 .774108-65. YES

    file = open(path_of_pch_file, 'r')
    file_b = open(path_of_new_bdf_file, 'w')
    keep_line = False
    keep_continuation_line = False

    # Example: Suppose you only want to read PCOMP entries
    # 1 PCOMP 10000001-.0105 0.0 650000. TSAI 0.0 0.0 SYM

    # 2 70 1.5 80. YES 70 .774108-65. YES
    # 3 70 1.5 80. YES 70 .774108-65. YES
    # 4 MAT1 101
    # 5 PSHELL* 10000001 110000001 9.09643308E+00 210000001*
    # 6 * 1.00000000E+00 0 1.00000000E+00 0.00000000E+00*
    # 7 * -1.05000000E-02 9.08593308E+00 410000001
    # 8 $MAT2* 110000001 1.92043294E+06 2.00203393E+06 -4.72326628E+05*
    # 9 $* 2.27776275E+07 3.50255102E+04 2.62296904E+06 5.85260000E-02*
    # 10 $* 0.00000000E+00 0.00000000E+00 0.00000000E+00 0.00000000E+00*
    # 11 $* 0.00000000E+00 0.00000000E+00 0.00000000E+00 0.00000000E+00*
    # 12 $* 0
    # 13 PCOMP 10000002-.0105 0.0 650000. TSAI 0.0 0.0 SYM
    # 14 70 1.5 80. YES 70 .774108-65. YES
    # 15 70 1.5 80. YES 70 .774108-65. YES

    for line in file:
        if re.match(r'^' + name_of_entry, line):
            # This detects lines 1 and 13 which is the first line of the entry
            keep_line = True
        elif re.match(r'^(\s+)', line) is None:
            # This detects lines 4, 5, 6, 7, 8, 9, 10, 11, 12 which are other entries not to keep
            keep_line = False
            keep_continuation_line = False
        if keep_line is True:
            if re.match(r'^(\s+)', line):
                # This detects lines 2, 3, 14, 15 which are continuation lines of the entry
                keep_continuation_line = True
            if re.match(r'^$', line) is None:
                # This detects all lines, except lines 8, 9, 10, 11, 12 which are commented with $
                if keep_line is True or keep_continuation_line is True:
                    # Write the line to a new file
                    file_b.write(line)

    file.close()
    file_b.close()

if __name__ == '__main__':
    # Comments
    # 1. This python script outputs updated PCOMP and CQUAD4 elements after an MSC Nastran topometry optimization.
    # 2. Modify path_a and path_b, then run this script
    # 3. This works as long MDLPRM,HDF5,1 is used, which triggers the output of the
    # INPUT datasets to the H5 file. The INPUT datasets are the bulk data entries: GRIDs, CQUAD4s, PSHELLs,
    etc.

    path_a = '/home/usera/Downloads/nastran_working_directory/model.h5'
    path_b = '/home/usera/Downloads/nastran_working_directory/model.pch'

    # Output New QUAD4 Elements After Topometry Optimization
    # #####
    # Output updated CQUAD4 entries
    read_cquad4_entries_from_h5_and_write_to_bdf(path_a, 'new_2D_elements.tmp')

    # Output
    # CQUAD4*,1,1000000001,1,2,16,15
    # CQUAD4*,2,1000000002,2,3,17,16
    # CQUAD4*,3,1000000003,3,4,18,17
    # CQUAD4*,4,1000000004,4,5,19,18
    # CQUAD4*,5,1000000005,5,6,20,19
    # CQUAD4*,6,1000000006,6,7,21,20
    # CQUAD4*,7,1000000007,7,8,22,21
    # [...]

    # Output New PCOMP Entries After Topometry Optimization
    # #####
    filter_entries_from_pch(path_b, 'PCOMP', 'new_pcomp_entries.tmp')
    # Outputs
    # PCOMP 10000001-.0105 0.0 650000. TSAI 0.0 0.0 SYM
    # 70 1.5 80. YES 70 .774108-65. YES
    # 70 1.5 80. YES 70 .774108-65. YES
    # PCOMP 10000002-.0105 0.0 650000. TSAI 0.0 0.0 SYM
    # 70 1.39312 80. YES 70 .052964-65. YES
    # 70 1.39312 80. YES 70 .052964-65. YES
```

Update the Original Model

The Python script generates a new TMP file.

1. Copy and paste the CQUAD4 elements to the original BDF file.

new_2D_elements.tmp

1	CQUAD4,1,10000001,1,2,23,22
2	CQUAD4,2,10000002,2,3,24,23
3	CQUAD4,3,10000003,3,4,25,24
4	CQUAD4,4,10000004,4,5,26,25
5	CQUAD4,5,10000005,5,6,27,26
6	CQUAD4,6,10000006,6,7,28,27
7	CQUAD4,7,10000007,7,8,29,28
8	CQUAD4,8,10000008,8,9,30,29
9	CQUAD4,9,10000009,9,10,31,30
10	CQUAD4,10,10000010,10,11,32,31
11	CQUAD4,11,10000011,11,12,33,32
12	CQUAD4,12,10000012,12,13,34,33
13	CQUAD4,13,10000013,13,14,35,34
14	CQUAD4,14,10000014,14,15,36,35
15	CQUAD4,15,10000015,15,16,37,36
16	CQUAD4,16,10000016,16,17,38,37
17	CQUAD4,17,10000017,17,18,39,38
18	CQUAD4,18,10000018,18,19,40,39
19	CQUAD4,19,10000019,19,20,41,40
20	CQUAD4,20,10000020,20,21,42,41
21	CQUAD4,21,10000021,22,23,44,43
22	CQUAD4,22,10000022,23,24,45,44
23	CQUAD4,23,10000023,24,25,46,45
24	CQUAD4,24,10000024,25,26,47,46
25	CQUAD4,25,10000025,26,27,48,47
26	CQUAD4,26,10000026,27,28,49,48
27	CQUAD4,27,10000027,28,29,50,49
28	CQUAD4,28,10000028,29,30,51,50
29	CQUAD4,29,10000029,30,31,52,51
30	CQUAD4,30,10000030,31,32,53,52
31	CQUAD4,31,10000031,32,33,54,53
32	CQUAD4,32,10000032,33,34,55,54
33	CQUAD4,33,10000033,34,35,56,55
34	CQUAD4,34,10000034,35,36,57,56

Original BDF File

24	CQUAD4	1	1	1	2	23	22
25	CQUAD4	2	1	2	3	24	23
26	CQUAD4	3	1	3	4	25	24
27	CQUAD4	4	1	4	5	26	25
28	CQUAD4	5	1	5	6	27	26
29	CQUAD4	6	1	6	7	28	27
30	CQUAD4	7	1	7	8	29	28
31	CQUAD4	8	1	8	9	30	29
32	CQUAD4	9	1	9	10	31	30
33	CQUAD4	10	1	10	11	32	31
34	CQUAD4	11	1	11	12	33	32
35	CQUAD4	12	1	12	13	34	33
36	CQUAD4	13	1	13	14	35	34
37	CQUAD4	14	1	14	15	36	35
38	CQUAD4	15	1	15	16	37	36
39	CQUAD4	16	1	16	17	38	37
40	CQUAD4	17	1	17	18	39	38
41	CQUAD4	18	1	18	19	40	39
42	CQUAD4	19	1	19	20	41	40
43	CQUAD4	20	1	20	21	42	41
44	CQUAD4	21	1	22	23	44	43
45	CQUAD4	22	1	23	24	45	44
46	CQUAD4	23	1	24	25	46	45
47	CQUAD4	24	1	25	26	47	46
48	CQUAD4	25	1	26	27	48	47
49	CQUAD4	26	1	27	28	49	48
50	CQUAD4	27	1	28	29	50	49
51	CQUAD4	28	1	29	30	51	50
52	CQUAD4	29	1	30	31	52	51
53	CQUAD4	30	1	31	32	53	52
54	CQUAD4	31	1	32	33	54	53
55	CQUAD4	32	1	33	34	55	54
56	CQUAD4	33	1	34	35	56	55
57	CQUAD4	34	1	35	36	57	56

1

The Python script generates a new TMP file.

1. Copy and paste the PCOMP entries to the original BDF file.

```
new_pcomp_entries.tmp
```

Original BDF File

new_bcomp_entries.tmp

1

PCOMP

10000001-

.0105

0.0

650000.

TSAI

0.0

0.0

SYM

2

70

1.5

80.

YES

70

.774108-

65.

YES

3

70

1.5

80.

YES

70

.774108-

65.

YES

4

PCOMP

10000002-

.0105

0.0

650000.

TSAI

0.0

0.0

SYM

5

70

1.39312

80.

YES

70

.052964-

65.

YES

6

70

1.39312

80.

YES

70

.052964-

65.

YES

7

PCOMP

10000003-

.0105

0.0

650000.

TSAI

0.0

0.0

SYM

8

70

.946061

80.

YES

70

.011974-

65.

YES

9

70

.946061

80.

YES

70

.011974-

65.

YES

10

PCOMP

10000004-

.0105

0.0

650000.

TSAI

0.0

0.0

SYM

11

70

.713549

80.

YES

70

.001135-

65.

YES

12

70

.713549

80.

YES

70

.001135-

65.

YES

13

PCOMP

10000005-

.0105

0.0

650000.

TSAI

0.0

0.0

SYM

14

70

.534361

80.

YES

70

.001135-

65.

YES

15

70

.534361

80.

YES

70

.001135-

65.

YES

16

PCOMP

10000006-

.0105

0.0

650000.

TSAI

0.0

0.0

SYM

17

70

.399395

80.

YES

70

.001142-

65.

YES

18

70

.399395

80.

YES

70

.001142-

65.

YES

19

PCOMP

10000007-

.0105

0.0

650000.

TSAI

0.0

0.0

SYM

20

70

.317016

80.

YES

70

.00312

-65.

YES

21

70

.317016

80.

YES

70

.00312

-65.

YES

22

PCOMP

10000008-

.0105

0.0

650000.

TSAI

0.0

0.0

SYM

23

70

.25365

80.

YES

70

.015178-

65.

YES

24

70

.25365

80.

YES

70

.015178-

65.

YES

25

PCOMP

10000009-

.0105

0.0

650000.

TSAI

0.0

0.0

SYM

26

70

.205456

80.

YES

70

.020367-

65.

YES

27

70

.205456

80.

YES

70

.020367-

65.

YES

28

PCOMP

10000010-

.0105

0.0

650000.

TSAI

0.0

0.0

SYM

29

70

.161272

80.

YES

70

.021622-

65.

YES

30

70

.161272

80.

YES

70

.021622-

65.

YES

31

PCOMP

10000011-

.0105

0.0

650000.

TSAI

0.0

0.0

SYM

32

70

.116866

80.

YES

70

.055676-

65.

YES

tomex500.dat

1364

SPC1

1

123456

1

THRU

21

1365

LOAD

3

1.

1.

1

1366

\$ Nodal Forces of Load Set : force

1367

FORCE

1

683

0

300.

1.

-1.

0.

1368

\$ Referenced Coordinate Frames

1369

mat1

1

2.07+5

.3

7.93

1370

PCOMP

1

-.0105

0.0

0.65E6

TSAI

SYM

1371

70

1.000

80.

YES

70

1.000

-65.

YES

YES

1372

70

1.000

80.

YES

70

1.000

-65.

YES

YES

1373

\$ 1 || 2 || 3 || 4 || 5 || 6 || 7 || 8 || 9 ||

1374

\$ Composite Type: T300/5208 (Graphite/Expoxy)

1375

\$

1376

\$ E1: 26.25E6 PSI

1377

\$ E2: 1.49E6 PSI

1378

\$ v12: .28

1379

\$ G12: 1.04E6 PSI

1380

\$

1381

\$ Density: 5.8526E-2 lbm/in^3

1382

\$

1383

\$ Ultimate longitudinal tensile strength: 217.56E3 PSI

1384

\$ Ultimate longitudinal compressive strength: 217.56E3 PSI

1385

\$ Ultimate transverse tensile strength: 5.802E3 PSI

1386

\$ Ultimate transverse compressive strength: 35.68E3 PSI

1387

\$ Ultimate in-plane shear strength: 9.863E3 PSI

1388

\$

1389

\$ Source: Tsai, S.W. and Hahn, H.T., Introduction to Composite Materials,

1390

\$ FL, Table 1.7, p. 19; Table 7.1, p.292; Table 8.3, p. 342.

1391

\$ 1 || 2 || 3 || 4 || 5 || 6 || 7 || 8 || 9 ||

1392

MAT8

70

26.25e6

1.49e6

.28

1.04e6

.058526

1393

217560.

217560.

5802.

35683.

9863.

1394

ENDDATA

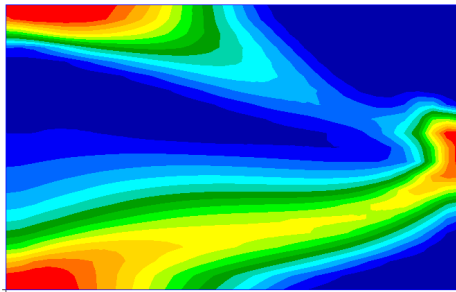
5c8b7a30

1395

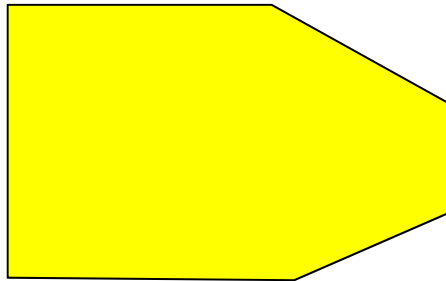
1

Results

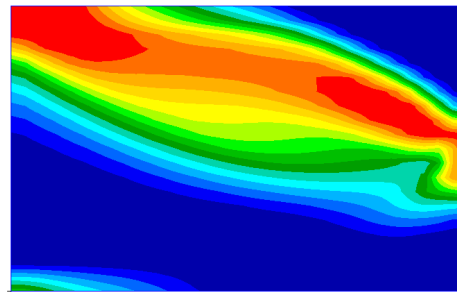
Topometry
Results



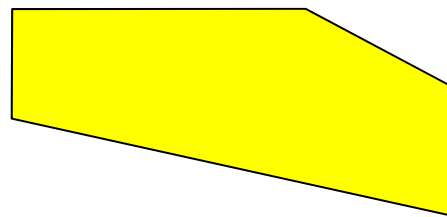
Ply 1 and 8
80°



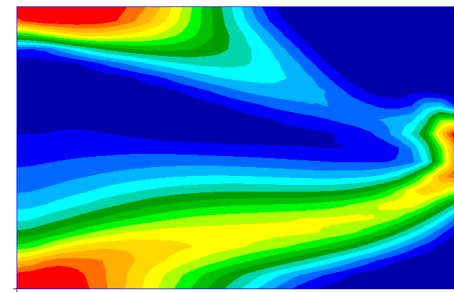
Ply 1 and 8
80°



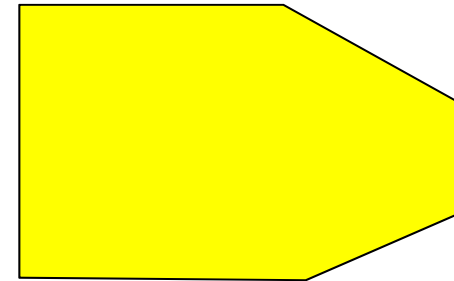
Ply 2 and 7
-65°



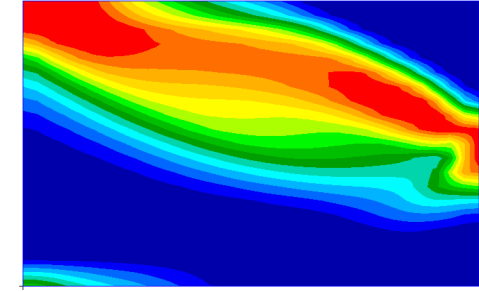
Ply 2 and 7
-65°



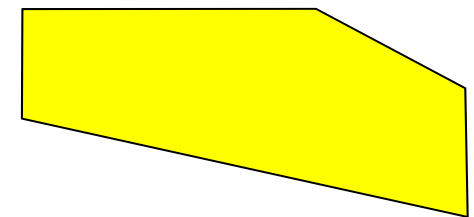
Ply 3 and 6
80°



Ply 3 and 6
80°



Ply 4 and 5
-65°



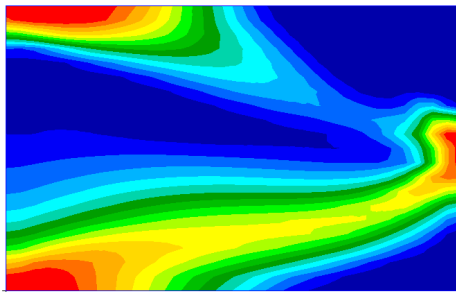
Ply 4 and 5
-65°

Possible Ply
Shapes

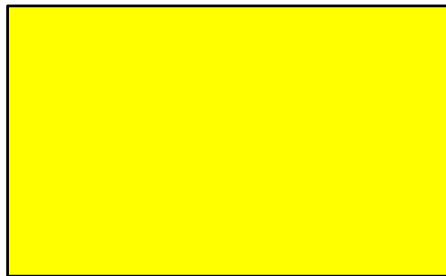
Results

Ply drop offs are employed for the -65 degree plies. In the interest of manufacturability, ply drop offs are not employed for the 80 degree plies.

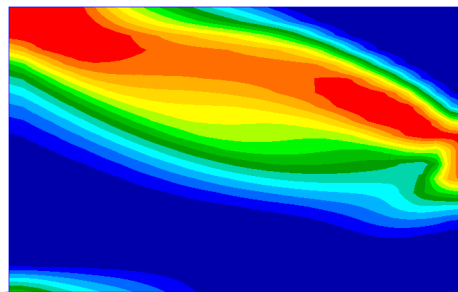
Topometry
Results



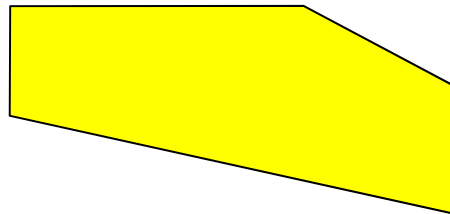
Ply 1 and 8
80°



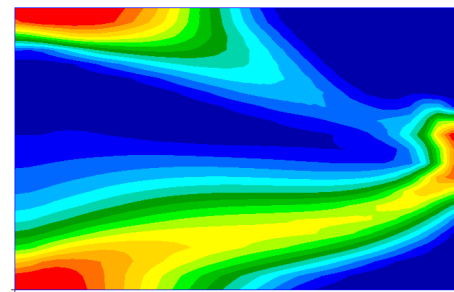
Ply 1 and 8
80°



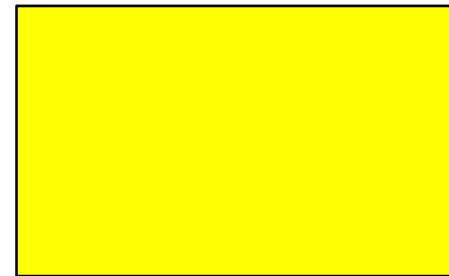
Ply 2 and 7
-65°



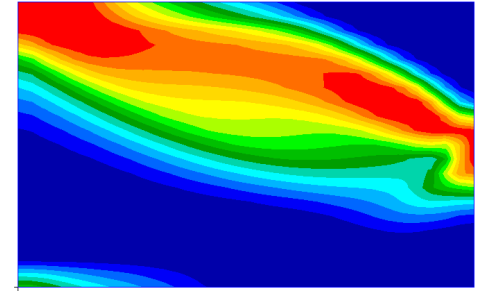
Ply 2 and 7
-65°



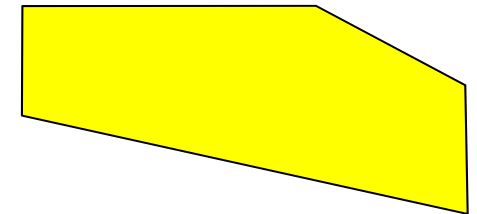
Ply 3 and 6
80°



Ply 3 and 6
80°



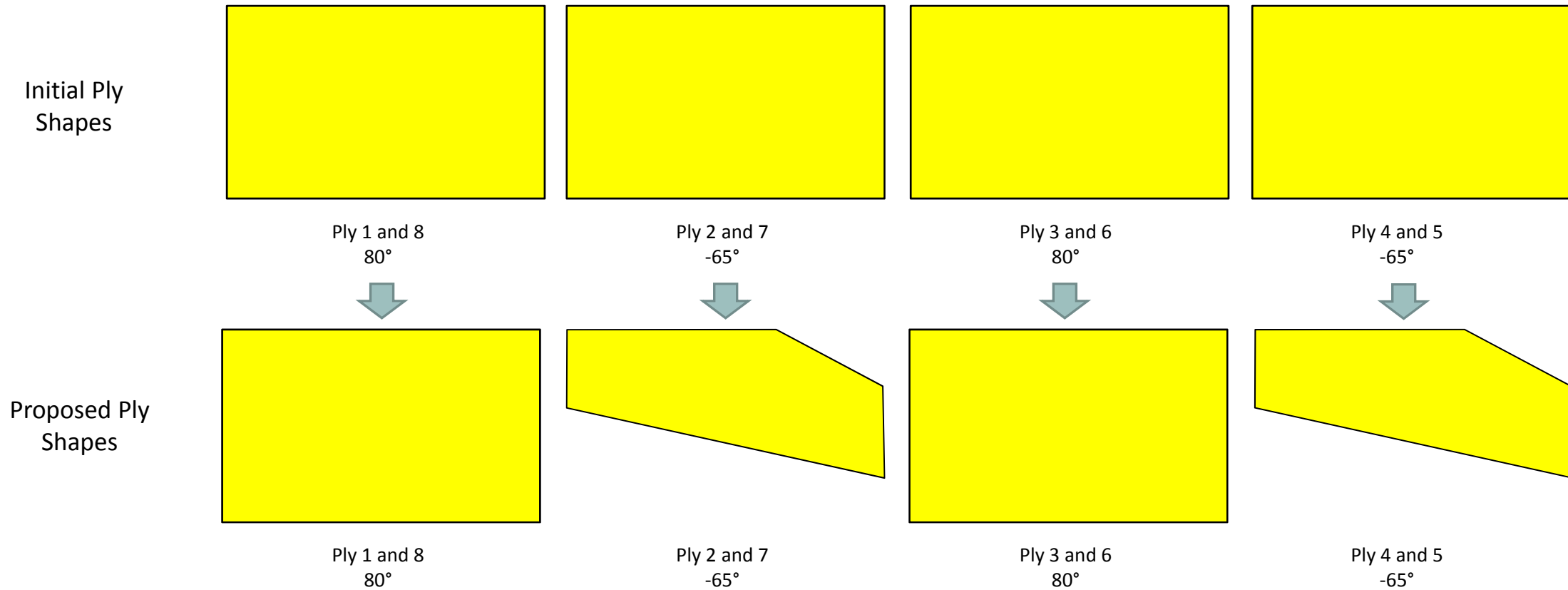
Ply 4 and 5
-65°



Ply 4 and 5
-65°

Proposed Ply
Shapes

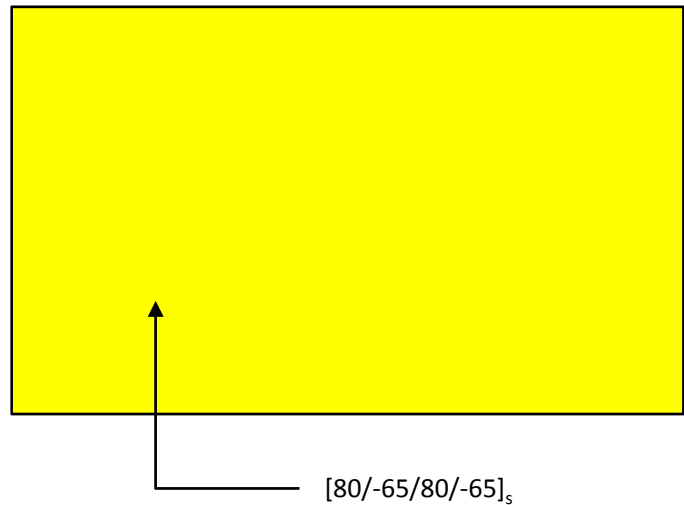
Results



Results

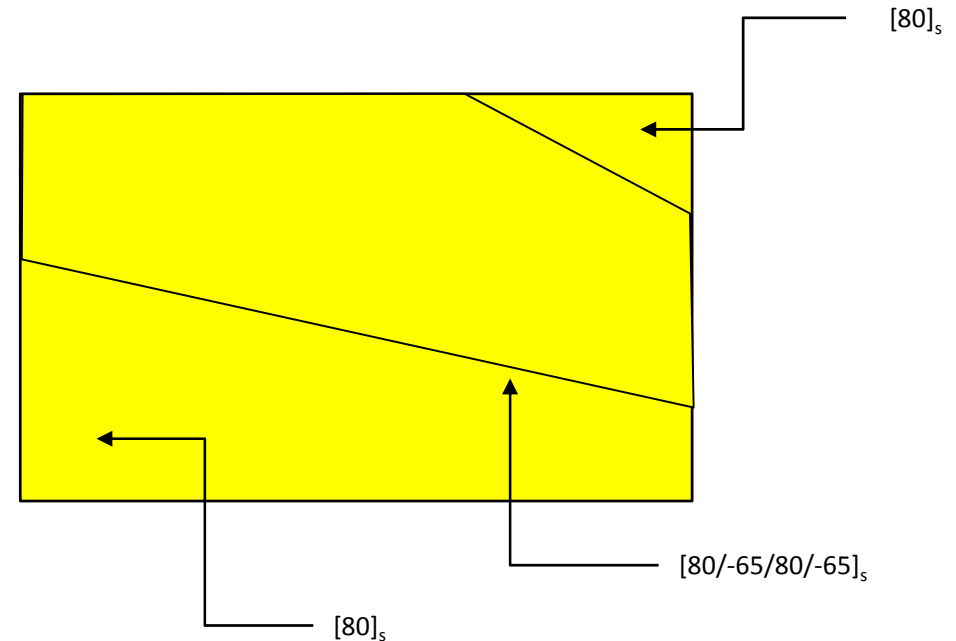
Before Optimization

- Panel of initial layup



After Optimization

- Panel of updated layups
- The weight is reduced



End of Tutorial